

Formula Sheet Physical Transport Phenomena (3T320), 2013/2014

Conservation laws of mass, momentum and energy in integral form

(for a control volume V , enclosed by a permeable surface S)

- Mass:
$$\frac{\partial}{\partial t} \iiint_V \rho dV = - \iint_S \rho(\vec{v} \cdot \vec{n}) dA$$

- Momentum:
$$\frac{\partial}{\partial t} \iiint_V \rho \vec{v} dV = - \iint_S \rho \vec{v}(\vec{v} \cdot \vec{n}) dA + \sum \vec{F}$$

$$\sum \vec{F} = - \iint_S p \vec{n} dA + \iint_S (\vec{n} \cdot \vec{\tau}) dA + \iiint_V \rho \vec{g} dV + \vec{F}_{\text{other}}$$

- Energy:
$$\frac{\partial}{\partial t} \iiint_V \rho e dV = - \iint_S \rho e(\vec{v} \cdot \vec{n}) dA + \frac{\delta Q}{dt} - \frac{\delta W}{dt}$$

$$e = \frac{v^2}{2} + gy + u; \quad W = W_s + W_p + W_\mu; \quad W_\mu = - \iint_S \vec{v} \cdot (\vec{n} \cdot \vec{\tau}) dA$$

$$\frac{\partial}{\partial t} \iiint_V \rho e dV = - \iint_S \left(e + \frac{p}{\rho} \right) \rho(\vec{v} \cdot \vec{n}) dA + \frac{\delta Q}{dt} - \frac{\delta W_s}{dt} - \frac{\delta W_\mu}{dt}$$

$$\frac{V_1^2}{2g} + y_1 + \frac{p_1}{\rho_1 g} = \frac{V_2^2}{2g} + y_2 + \frac{p_2}{\rho_2 g} + \sum h_{\text{turbine}} + \sum h_{\text{pump}} + h_L \quad (\text{head form})$$

$$\frac{V_1^2}{2g} + y_1 + \frac{p_1}{\rho g} = \frac{V_2^2}{2g} + y_2 + \frac{p_2}{\rho g} \quad (\text{Bernoulli's equation})$$

Stress tensor

$$\vec{\sigma} = -p\vec{I} + \vec{\tau}$$

Shear stresses for a laminar Newtonian fluid

- Cartesian coordinates (x, y, z) with $\vec{v} = (v_x, v_y, v_z)$:

$$\tau_{xy} = \mu \left(\frac{\partial v_x}{\partial y} + \frac{\partial v_y}{\partial x} \right); \quad \tau_{yz} = \mu \left(\frac{\partial v_y}{\partial z} + \frac{\partial v_z}{\partial y} \right); \quad \tau_{zx} = \mu \left(\frac{\partial v_z}{\partial x} + \frac{\partial v_x}{\partial z} \right)$$

- Cylindrical coordinates (r, θ, z) with $\vec{v} = (v_r, v_\theta, v_z)$:

$$\tau_{r\theta} = \mu \left(r \frac{\partial v_r}{\partial \theta} + r \frac{\partial}{\partial r} \left(\frac{v_\theta}{r} \right) \right); \quad \tau_{\theta z} = \mu \left(\frac{\partial v_\theta}{\partial z} + \frac{1}{r} \frac{\partial v_z}{\partial \theta} \right); \quad \tau_{zr} = \mu \left(\frac{\partial v_z}{\partial r} + \frac{\partial v_r}{\partial z} \right)$$

Normal stresses for a laminar Newtonian fluid

- Cartesian coordinates (x, y, z) with $\vec{v} = (v_x, v_y, v_z)$:

$$\tau_{xx} = \mu \left(2 \frac{\partial v_x}{\partial x} - \frac{2}{3} \vec{\nabla} \cdot \vec{v} \right); \quad \tau_{yy} = \mu \left(2 \frac{\partial v_y}{\partial y} - \frac{2}{3} \vec{\nabla} \cdot \vec{v} \right); \quad \tau_{zz} = \mu \left(2 \frac{\partial v_z}{\partial z} - \frac{2}{3} \vec{\nabla} \cdot \vec{v} \right)$$

- Cylindrical coordinates (r, θ, z) with $\vec{v} = (v_r, v_\theta, v_z)$:

$$\tau_{rr} = \mu \left(2 \frac{\partial v_r}{\partial r} - \frac{2}{3} \vec{\nabla} \cdot \vec{v} \right); \quad \tau_{\theta\theta} = \mu \left(2 \frac{1}{r} \frac{\partial v_\theta}{\partial \theta} - \frac{2}{3} \vec{\nabla} \cdot \vec{v} \right); \quad \tau_{zz} = \mu \left(2 \frac{\partial v_z}{\partial z} - \frac{2}{3} \vec{\nabla} \cdot \vec{v} \right)$$

Conservation laws of mass and momentum in differential form

- Mass: $\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \vec{v}) = 0$ or $\frac{D\rho}{Dt} + \rho \nabla \cdot \vec{v} = 0$
- Momentum: $\rho \frac{D\vec{v}}{Dt} = -\nabla p + \nabla \cdot \vec{\tau} + \rho \vec{g}$

Continuity equation for incompressible flow

- Vector form: $\nabla \cdot \vec{v} = 0$
- Cartesian coordinates (x, y, z) with $\vec{v} = (v_x, v_y, v_z)$:

$$\frac{\partial v_x}{\partial x} + \frac{\partial v_y}{\partial y} + \frac{\partial v_z}{\partial z} = 0$$

- Cylindrical coordinates (r, θ, z) with $\vec{v} = (v_r, v_\theta, v_z)$:

$$\frac{1}{r} \frac{\partial}{\partial r} (r v_r) + \frac{1}{r} \frac{\partial v_\theta}{\partial \theta} + \frac{\partial v_z}{\partial z} = 0$$

Navier-Stokes equations

(incompressible flow, Newtonian fluid, constant μ)

- Vector form:

$$\rho \frac{D\vec{v}}{Dt} = -\nabla p + \mu \nabla^2 \vec{v} + \rho \vec{g}$$

- Cartesian coordinates (x, y, z) with $\vec{v} = (v_x, v_y, v_z)$:

$$\rho \left(\frac{\partial v_x}{\partial t} + v_x \frac{\partial v_x}{\partial x} + v_y \frac{\partial v_x}{\partial y} + v_z \frac{\partial v_x}{\partial z} \right) = -\frac{\partial p}{\partial x} + \mu \left(\frac{\partial^2 v_x}{\partial x^2} + \frac{\partial^2 v_x}{\partial y^2} + \frac{\partial^2 v_x}{\partial z^2} \right) + \rho g_x$$

$$\rho \left(\frac{\partial v_y}{\partial t} + v_x \frac{\partial v_y}{\partial x} + v_y \frac{\partial v_y}{\partial y} + v_z \frac{\partial v_y}{\partial z} \right) = -\frac{\partial p}{\partial y} + \mu \left(\frac{\partial^2 v_y}{\partial x^2} + \frac{\partial^2 v_y}{\partial y^2} + \frac{\partial^2 v_y}{\partial z^2} \right) + \rho g_y$$

$$\rho \left(\frac{\partial v_z}{\partial t} + v_x \frac{\partial v_z}{\partial x} + v_y \frac{\partial v_z}{\partial y} + v_z \frac{\partial v_z}{\partial z} \right) = -\frac{\partial p}{\partial z} + \mu \left(\frac{\partial^2 v_z}{\partial x^2} + \frac{\partial^2 v_z}{\partial y^2} + \frac{\partial^2 v_z}{\partial z^2} \right) + \rho g_z$$

- Cylindrical coordinates (r, θ, z) with $\vec{v} = (v_r, v_\theta, v_z)$:

$$\begin{aligned} & \rho \left(\frac{\partial v_r}{\partial t} + v_r \frac{\partial v_r}{\partial r} + \frac{v_\theta}{r} \frac{\partial v_r}{\partial \theta} - \frac{v_\theta^2}{r} + v_z \frac{\partial v_r}{\partial z} \right) \\ &= -\frac{\partial p}{\partial r} + \mu \left[\frac{\partial}{\partial r} \left(\frac{1}{r} \frac{\partial}{\partial r} (r v_r) \right) + \frac{1}{r^2} \frac{\partial^2 v_r}{\partial \theta^2} - \frac{2}{r^2} \frac{\partial v_\theta}{\partial \theta} + \frac{\partial^2 v_r}{\partial z^2} \right] + \rho g_r \\ & \rho \left(\frac{\partial v_\theta}{\partial t} + v_r \frac{\partial v_\theta}{\partial r} + \frac{v_\theta}{r} \frac{\partial v_\theta}{\partial \theta} + \frac{v_r v_\theta}{r} + v_z \frac{\partial v_\theta}{\partial z} \right) \\ &= -\frac{1}{r} \frac{\partial p}{\partial \theta} + \mu \left[\frac{\partial}{\partial r} \left(\frac{1}{r} \frac{\partial}{\partial r} (r v_\theta) \right) + \frac{1}{r^2} \frac{\partial^2 v_\theta}{\partial \theta^2} + \frac{2}{r^2} \frac{\partial v_r}{\partial \theta} + \frac{\partial^2 v_\theta}{\partial z^2} \right] + \rho g_\theta \\ & \rho \left(\frac{\partial v_z}{\partial t} + v_r \frac{\partial v_z}{\partial r} + \frac{v_\theta}{r} \frac{\partial v_z}{\partial \theta} + v_z \frac{\partial v_z}{\partial z} \right) \\ &= -\frac{\partial p}{\partial z} + \mu \left[\frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial v_z}{\partial r} \right) + \frac{1}{r^2} \frac{\partial^2 v_z}{\partial \theta^2} + \frac{\partial^2 v_z}{\partial z^2} \right] + \rho g_z \end{aligned}$$

Rate equations for heat transfer

(with $\dot{Q} = \delta Q / dt$)

- Fourier's law: $\frac{\dot{Q}}{A} = -k \nabla T$
- Newton's 'ansatz': $\frac{\dot{Q}}{A} = h \Delta T$
- Stefan-Boltzmann law: $\frac{\dot{Q}}{A} = \sigma T^4$
- Overall heat transfer: $\frac{\dot{Q}}{A} = \frac{\Delta T}{AR_{\text{thermal}}} = U \Delta T$

Integral form of the energy equation without friction and shaft power

$$\iiint_V \dot{q} dV + \dot{Q} = \frac{\partial}{\partial t} \iiint_V \rho e dV + \iint_S \left(e + \frac{p}{\rho} \right) \rho (\vec{v} \cdot \vec{n}) dA$$

Differential form of the energy equation for incompressible media without friction

$$\dot{q} + \nabla \cdot (k \nabla T) = \rho c \frac{DT}{Dt}$$

Dimensionless parameters

$$\begin{aligned} \text{Re} &= \frac{\rho V L}{\mu}, & \text{Eu} &= \frac{p}{\rho V^2}, & \text{Fr} &= \frac{V^2}{gL}, & C_D &= \frac{F/A}{\frac{1}{2} \rho V^2} \\ \text{Fo} &= \frac{\alpha t}{(V/A)^2}, & \text{Bi} &= \frac{h(V/A)}{k}, & \text{Pr} &= \frac{\nu}{\alpha}, & \text{Nu}_L &= \frac{hL}{k} \end{aligned}$$

**EINDHOVEN UNIVERSITY OF TECHNOLOGY
DEPARTMENT OF APPLIED PHYSICS
FLUID DYNAMICS LABORATORY**

**Examination Physical Transport Phenomena (3T320),
Tuesday 28 January 2014, 14.00-17.00 hours.**

This example examination is given a maximum of 30 credit points. The maximum number of credit points per subtask is denoted between the brackets.

Problem 1

Answer the following questions with 'yes' or 'no' and give a short argumentation. No credit points will be given for answers without argumentation, even if the answer is right.

(1 pnt) (a) Consider a two-dimensional (dimensionless) velocity field $\vec{v} = (v_x, v_y) = (-ax + ay^2, ay)$, with a a constant. Is this flow incompressible?

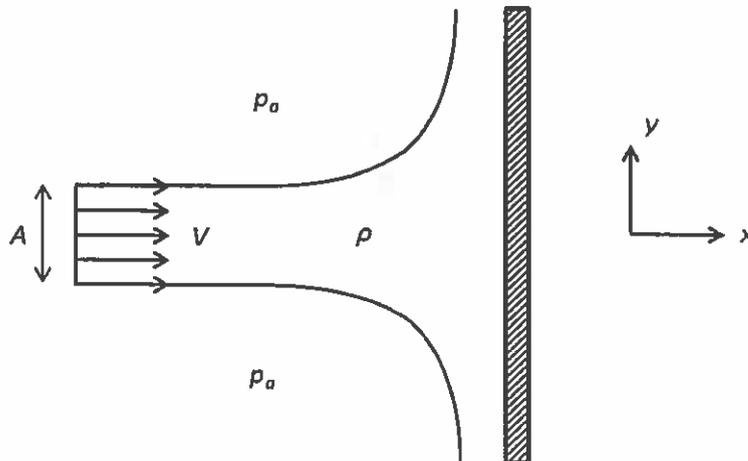
(1 pnt) (b) Is it true that the convective acceleration of a tracer at (x, y) in the flow field defined in (a) is given by $\frac{D\vec{v}}{Dt} = (\frac{Dv_x}{Dt}, \frac{Dv_y}{Dt}) = (a^2x + a^2y^2, a^2y)$?

(1 pnt) (c) Is the viscous stress tensor for the flow field defined in (a) given by

$$2\mu \begin{pmatrix} -a & ay \\ ay & a \end{pmatrix} ?$$

(1 pnt) (d) May the Bernoulli equation be applied in a laminar boundary-layer flow?

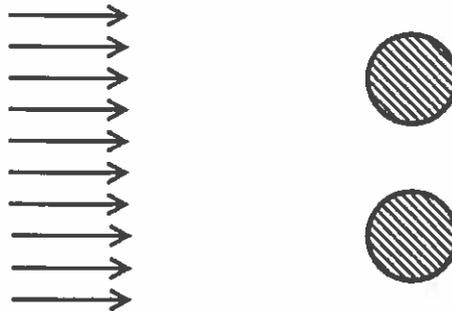
(1 pnt) (e) A steady, incompressible, frictionless, jet of fluid impinges perpendicularly on a flat plate. The jet has a cross-sectional area A , uniform velocity V , uniform density ρ , and is directed in the positive x -direction. The jet is adapted to the ambient pressure p_a . Gravitational forces are to be neglected. Is the force to hold the plate steady given by $\vec{F} = (F_x, F_y, F_z) = (-\rho AV^2, 0, 0)$?



- (1 pt) (f) Consider a nuclear explosion in which a large amount of energy E is suddenly released in the air. Experimental evidence suggests that the distance R travelled by the nuclear-explosion shock wave depends on time t as well as the energy E and the density of the ambient air, ρ . Is it true that this flow problem can be expressed in terms of a single dimensionless parameter

$$\Pi = \frac{R}{E^{1/5} t^{2/5} \rho^{-1/5}} ?$$

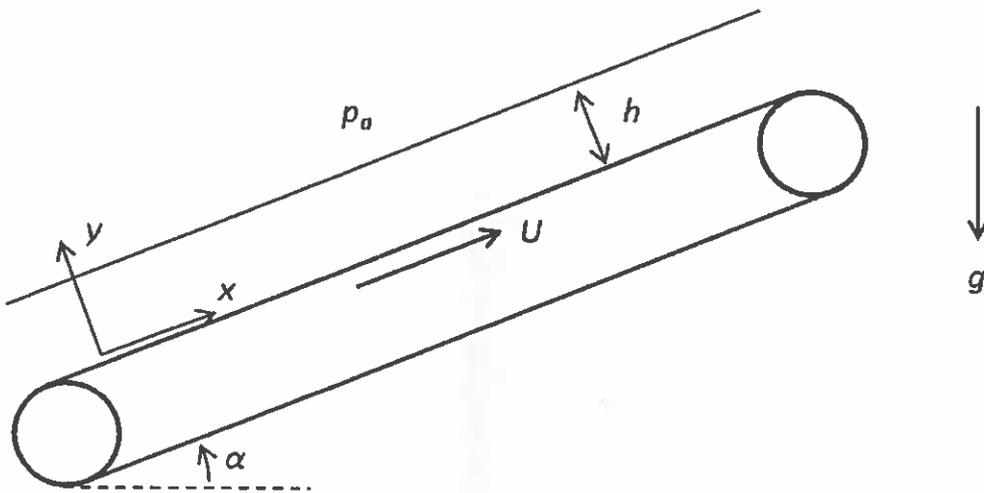
- (1 pt) (g) Consider an inviscid flow around two identical cylinders as sketched below. Is it true that the flow will push the cylinders apart from each other?



- (1 pt) (h) Do the Biot number and the Nusselt number have the same dimensions?
- (1 pt) (i) We immerse a boiled egg into cold water. The Biot Number is 1. Can we model the temperature-time dependence of the egg according to a lumped-parameter model?
- (1 pt) (j) For laminar flow past a flat surface, the Prandtl Number equals 50. Can Blasius' solution be used to model the thermal boundary layer?

Problem 2

Consider a transport belt that is used to transport a Newtonian incompressible fluid upwards. The transport belt is inclined by an angle α with respect to the horizontal, see sketch below. The upper side of the transport belt moves with a constant velocity U in the direction indicated in the sketch. On top of the belt is a layer of fluid with a constant thickness h . The fluid flow is steady and can be considered fully developed. The length L and width W of the belt are very large, such that boundary effects can be neglected. The flow can be considered two-dimensional (in the plane of the drawing). The Cartesian axes are chosen so that the x -axis is parallel to the belt and the y -axis is normal to the belt. The dynamic viscosity μ and the density ρ are constant. The stress force exerted by the air on the free surface of the fluid layer may be neglected, and the constant pressure of the environment is equal to p_a . The orientation of the gravitational acceleration is indicated in the figure. The components of the velocity in liquid film in the x - and y -direction are named v_x and v_y , respectively.



- (1 pnt) (a) State the boundary condition for the velocity at the upper side of the transport belt ($y = 0$).
- (1 pnt) (b) Prove that everywhere in the fluid layer the velocity component normal to the transport belt is zero, i.e. $v_y = 0$.
- (1 pnt) (c) Show that the boundary condition for the velocity at the interface between fluid and air ($y = h$) is given by $dv_x/dy = 0$.
- (2 pnt) (d) Prove that the x - and y -components of the Navier–Stokes equation can be reduced as follows:

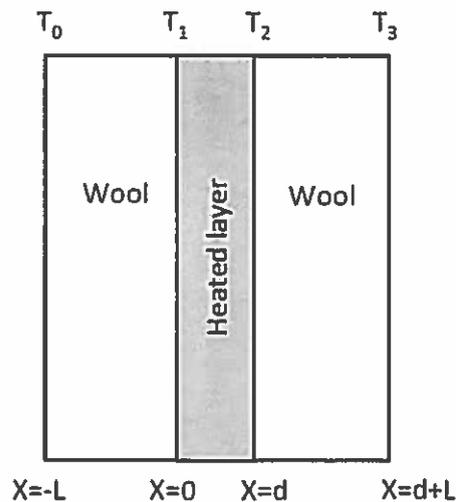
$$-\frac{\partial p}{\partial x} - \rho g \sin \alpha + \mu \frac{d^2 v_x}{dy^2} = 0 ,$$

$$-\frac{\partial p}{\partial y} - \rho g \cos \alpha = 0 .$$

- (1 pnt) (e) Use the y -component of the reduced Navier–Stokes equation to determine the pressure as a function of the height y in the fluid layer.
- (2 pnt) (f) Determine the velocity profile $v_x(y)$ and make a sketch of the solution.
- (1 pnt) (g) Determine the velocity of the belt, U , such that the mass flux in the fluid layer is zero.
- (1 pnt) (h) Compute the force that is exerted on the transport belt. Compute also the power needed to keep the belt running.

Problem 3

In a test set-up the thermal properties of electrical blankets are measured by clamping them in between two flat plates with fixed and applied temperatures T_0 and T_3 . The locations of these plates are at $x = -L$ and $x = d + L$. The geometry is sketched below. In the center of the blanket an electrically heated layer is present. It has a thickness d and thermal conductivity k_d . The heating per unit of volume is \dot{q} . The two wool insulation sections have a thickness L . The thermal conduction coefficient in the wool sections is k_w .



- (1 pnt) (a) State Fourier's law explicitly for the three sections of this problem: left wool layer, heated layer, and right wool layer, to compute the heat flux \dot{Q}/A in those layers.
- (2 pnt) (b) Assume first that the center layer is *not* heated. Find the relation that describes the heat flux \dot{Q}/A as a function of $T_0 - T_3$. Hint: use the expressions found in (a). Make a graph of the temperature as a function of x for $-L \leq x \leq L + d$. Note that $k_w < k_d$.
- (2 pnt) (c) Now in the heated layer an electric current is applied so there is heat generated

with a value \dot{q} per unit volume. The differential form of the steady-state energy equation is given by

$$k \frac{d^2 T}{dx^2} + \dot{q} = 0 .$$

Integrate this equation twice to find the temperature T in the heated layer as a function of x . Use as boundary conditions that at $x = 0$ (i) there is no heat flux ($dT/dx = 0$) and (ii) $T = T_1$. Compute the heat flux Q/A as a function of x in the heated layer.

- (2 pnt) (d) Find the expressions for the temperature distribution in the three layers by using the results of part c), and applying the correct boundary conditions. Give explicitly the relationship between \dot{q} and $T_0 - T_3$.
- (1 pnt) (e) Make a graph of the temperature as a function of x for $-L \leq x \leq L + d$. Is T_3 higher or lower than T_0 ?
- (2 pnt) (f) Under which condition can the temperature difference $T_1 - T_2$ be neglected with respect to the temperature difference $T_0 - T_3$?